

Free Space Optical Communications with Nanosatellites

K. Cahoy, MIT



CubeSats can enable persistent global imaging

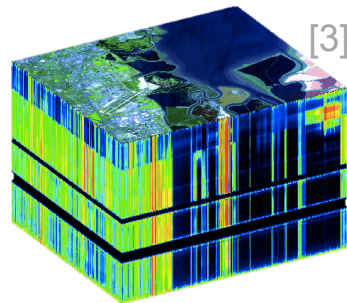
100 Gb =



80,000 images
~ 1 every second for a day



2 hours of GoPro video



120 Hyperspectral Image Cubes

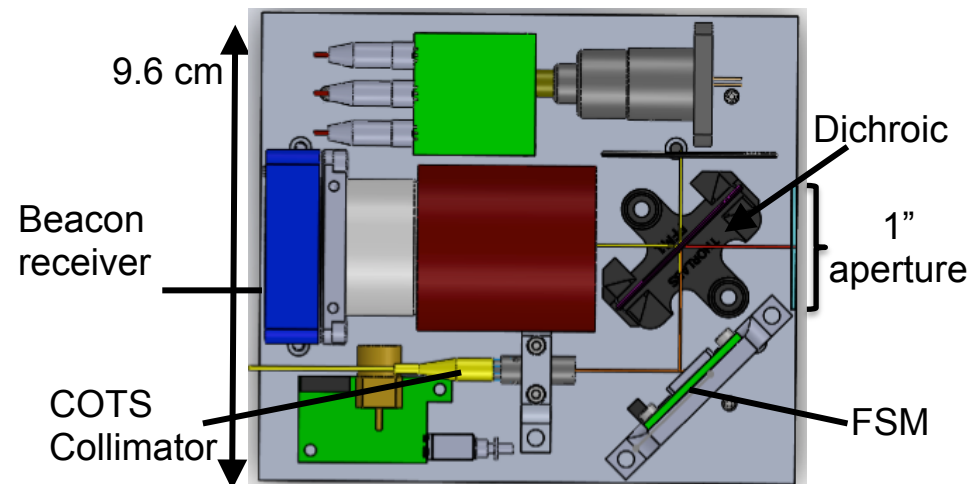
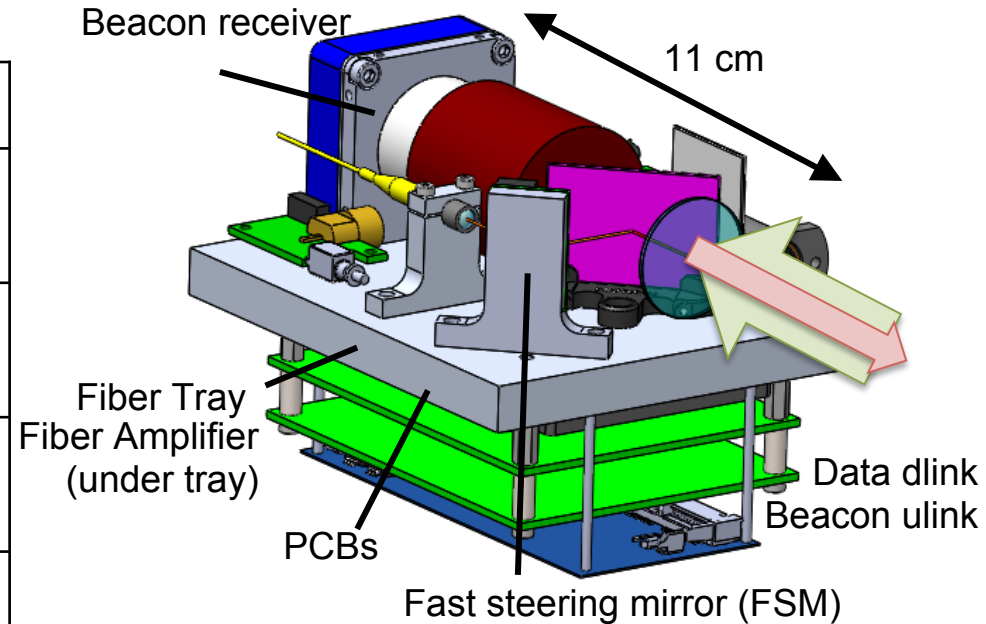
E. Clements, MIT

Low-cost spacecraft and access to space, but...
resource constraints limit data transfer

Downlink: NODE Space Terminal

NODE: Nanosatellite Optical Downlink Experiment

Scope	Low-cost CubeSat payload
Architecture	Direct detection MOPA COTS telecom parts (1550 nm)
Downlink data rates	10 Mbps (amateur telescope) 100 Mbps (OCTL)
Power	200 mW Tx < 15 W during Tx
Beamwidth	1.3 mrad HPBW (initial experiment)
Modulation	PPM
Coding	RS(255,239)
Mass, volume	1.0 kg, 1 U
Control architecture	<ul style="list-style-type: none"> • Bus coarse pointing (<math><0.5^\circ</math>) • FSM throw (<math>\pm 3^\circ</math>) • Beacon receiver (976 nm) for pointing knowledge (20 arcsec)



Downlink: NODE Ground Terminal

Downlink with NODE amateur telescope:

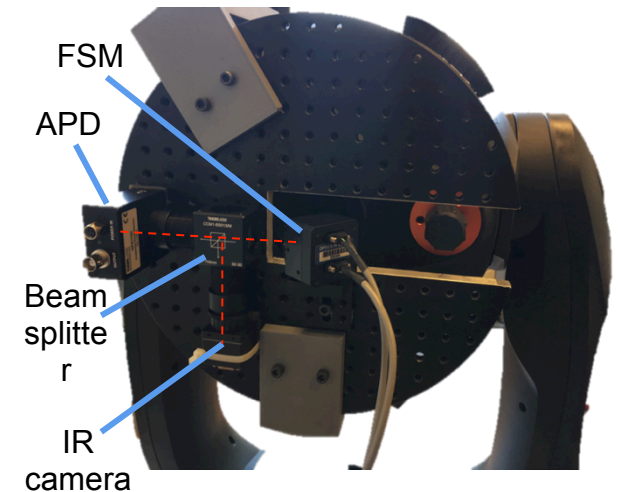
Data rate	10 - 50 Mbps
Rx Diameter	30 cm
Detector	Direct detection w/ COTS Voxel APD
Receiver electronics	Custom data acquisition system
Pointing	COTS IR camera and star tracker ^[19] FSM to keep spot on APD (no AO)
Uplink beacon	976 nm (OCTL) beacon ^[20]



E. Clements, K. Riesing

Downlink with JPL OCTL telescope:

Data rate	50 - 100 Mbps
Rx Diameter	1 m
Detector	Direct detection w/ COTS Voxel APD
Receiver electronics	Custom data acquisition system
Uplink beacon	976 nm, 10 W tx power, 1 mrad beam

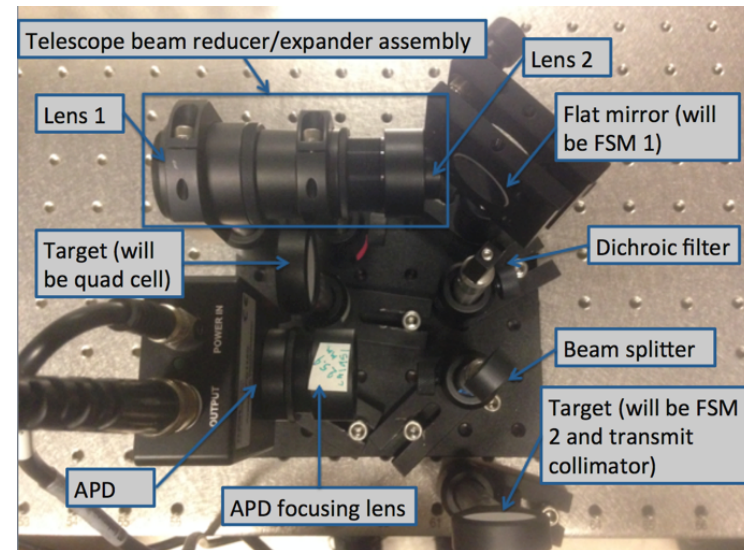
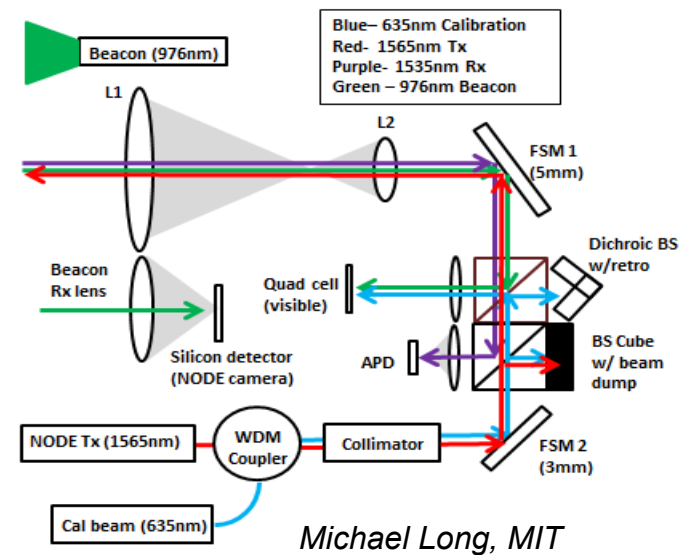


[4] Yoon, H., K. Riesing, and K. Cahoy. "Satellite Tracking System using Amateur Telescope and Star Camera for Portable Optical Ground Station." SmallSat 2016.

Crosslink: CLICK Space Terminal

CLICK: Cubesat Laser Intersatellite Crosslink

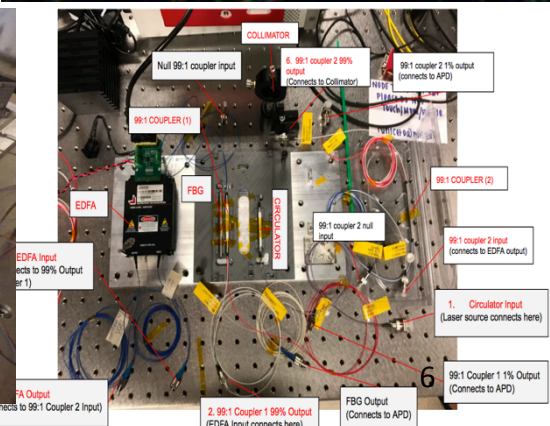
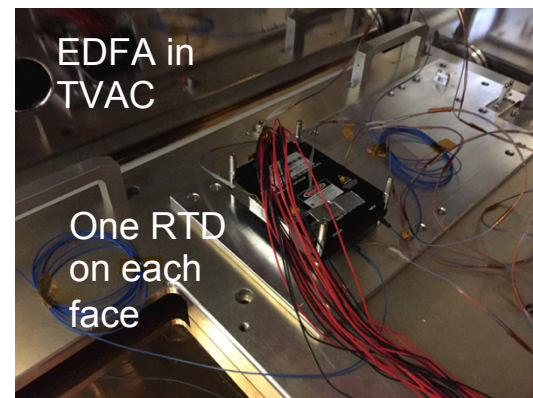
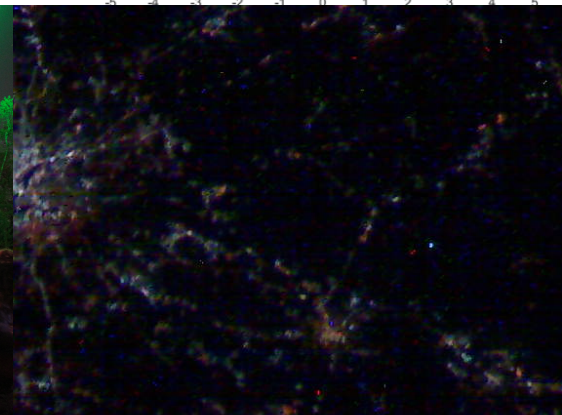
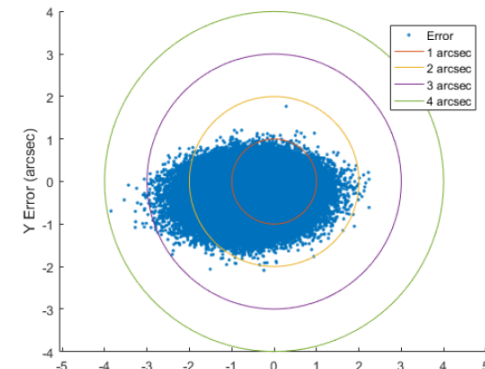
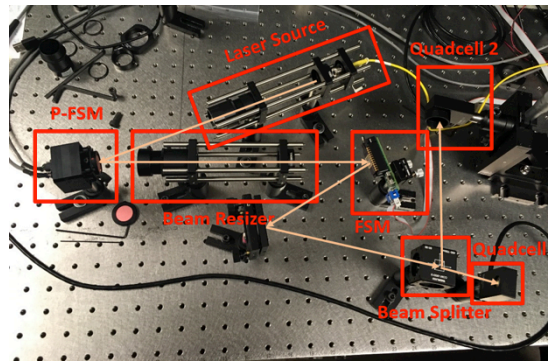
Scope	Monostatic, full-duplex
Architecture	Direct detection MOPA 1565 nm, 1535 nm tx/rx 976 nm beacon, 635 nm cal
Transmit divergence Transmit beam size	+/-0.07 mrad (FWHM) 16 mm (1/e ² Gaussian diam.)
Power	200 mW Tx , 500 mW beacon
Acq/Rx entrance aperture	20 mm
Telescope magnification	10.5x
Tx small beam size Acq/rx small beam size	1.6 mm 2.0 mm
Max dimension	106 mm
Quad cell FOV (acquisition) Quad cell linear range (acq)	+/-0.5° +/-0.15°
Beacon spot size on quad cell	1 mm
Data rate, range	20 Mbps at > 1000 km



Rachel Morgan, MIT

Recent MIT Accomplishments

- < 1 arcsec, 1-sigma pointing control w/ MEMS FSM vs. 40" 1-sigma CubeSat disturbances (H. Yoon, Ph.D. 2017)
- Internal MEMS FSM feedback calibration laser tracking beacon (O. Cierny, visiting MS 2017)
- TVac and radiation testing of EDFA
- Flight electronics boards and structures for downlink module
- Control software for alignment of and LEO tracking with COTS telescope
- Over the air testing of modulation, coding, and interleaving and clock recovery; also GS peak power tracking (Ziegler, SM 2017, Riesing Ph.D. 2018)
- Detailed design and simulation of monostatic crosslink terminal (Long, SM 2018, Morgan SB 2018)



Citations & Other Compact Space Lasercom



Citations:

[1] <https://www.planet.com/gallery/>

[2] <http://www.businessinsider.com/long-lost-gopro-footage-from-the-edge-of-space-2015-9>

[3] https://aviris.jpl.nasa.gov/data/image_cube.html, <https://www.nasa.gov/centers/dryden/research/AirSci/ER-2/aviris.html>

[4] Yoon, Hyosang, Kathleen Riesing, and Kerri Cahoy. "Satellite Tracking System using Amateur Telescope and Star Camera for Portable Optical Ground Station." (2016).

Other Compact Space Lasercom

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Analytical Space, <https://www.analyticalspace.com/>