
Overview on Lasercom (from an MIT-LL Perspective)

Scott A. Hamilton

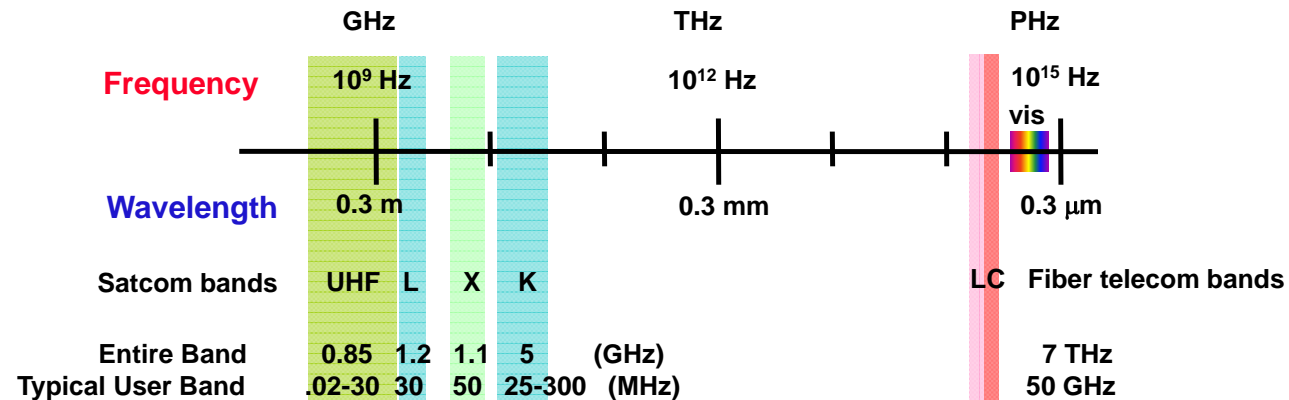
***Presented to:* Workshop on Free Space Optical Networks**

13-14 July 2017





Why Free-Space Optical Communications?



High Carrier Frequency

- Extremely Wide Bandwidth

Short Wavelength

- High Beam Directionality ($\sim \lambda/D$)

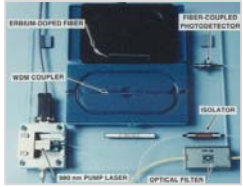
System benefits

- High data rates in *unregulated* bands
- Efficient power delivery for low SWaP
- Security through narrow beams (LPD/LPI)



MIT-LL Lasercom Flight Demonstrations

Technology Development (80's-90's)



High-power and Low-noise Optical Amps (pre-industry)

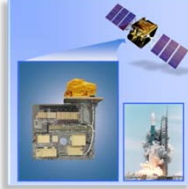


First Space-qualified >1 KHz Tracking Mirror



Lasercom Terminal Integration and Full-functional Testing

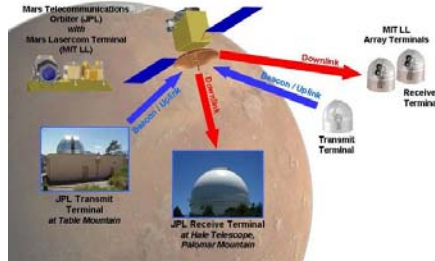
GeoLITE (2001)



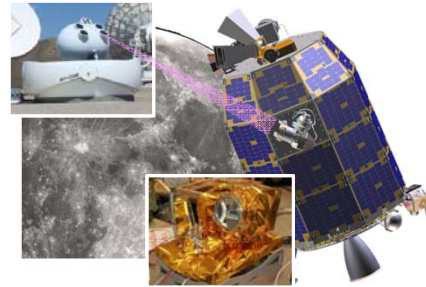
ALEX (2002)



Mars Lasercom Demo (2003-05)



Lunar Lasercom Demo (2008-14)



Lasercom Relay Demo (2011-19)



Free-space Optical Comm Airborne Link (2008-10)



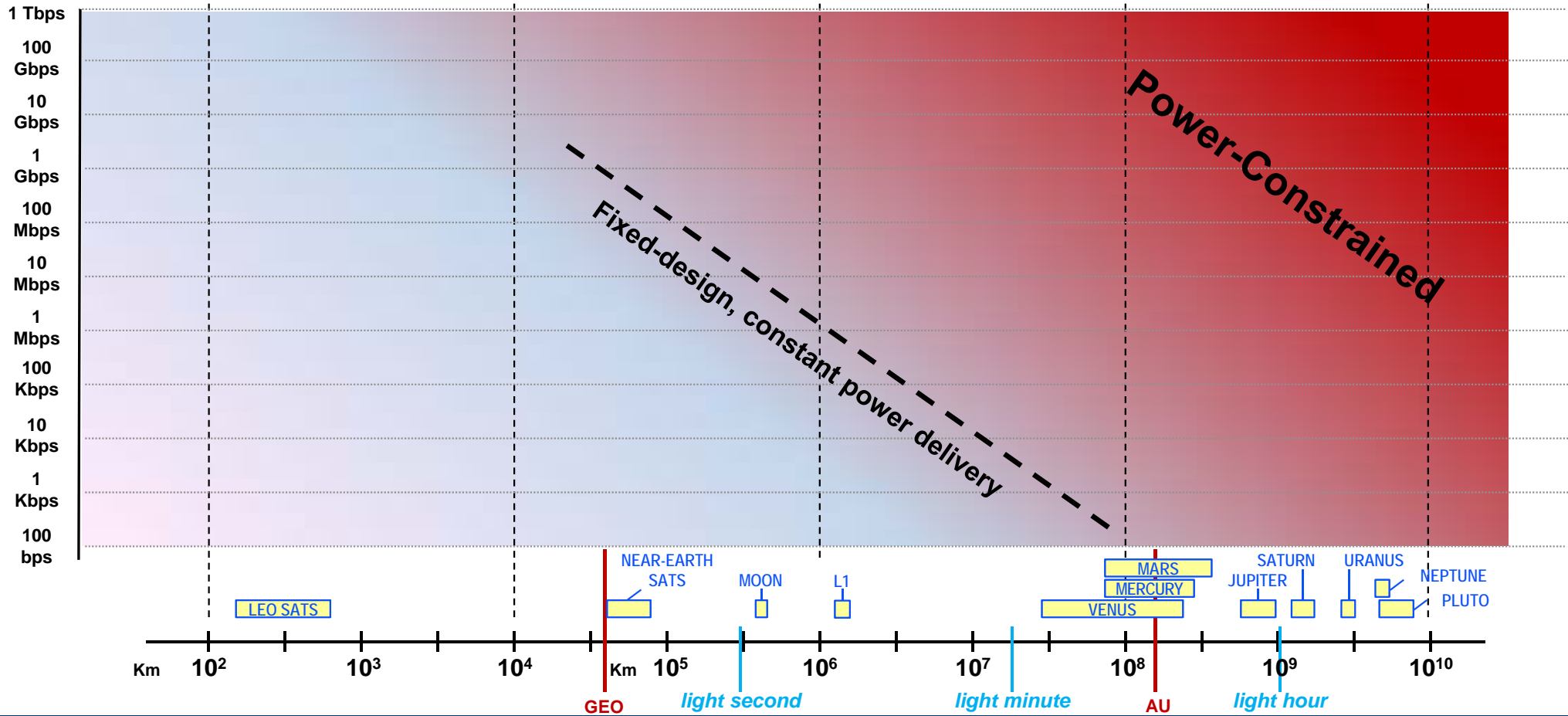
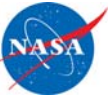
Compact Laser Transmitter (2009-14)



Perceived lasercom technical challenges have been solved

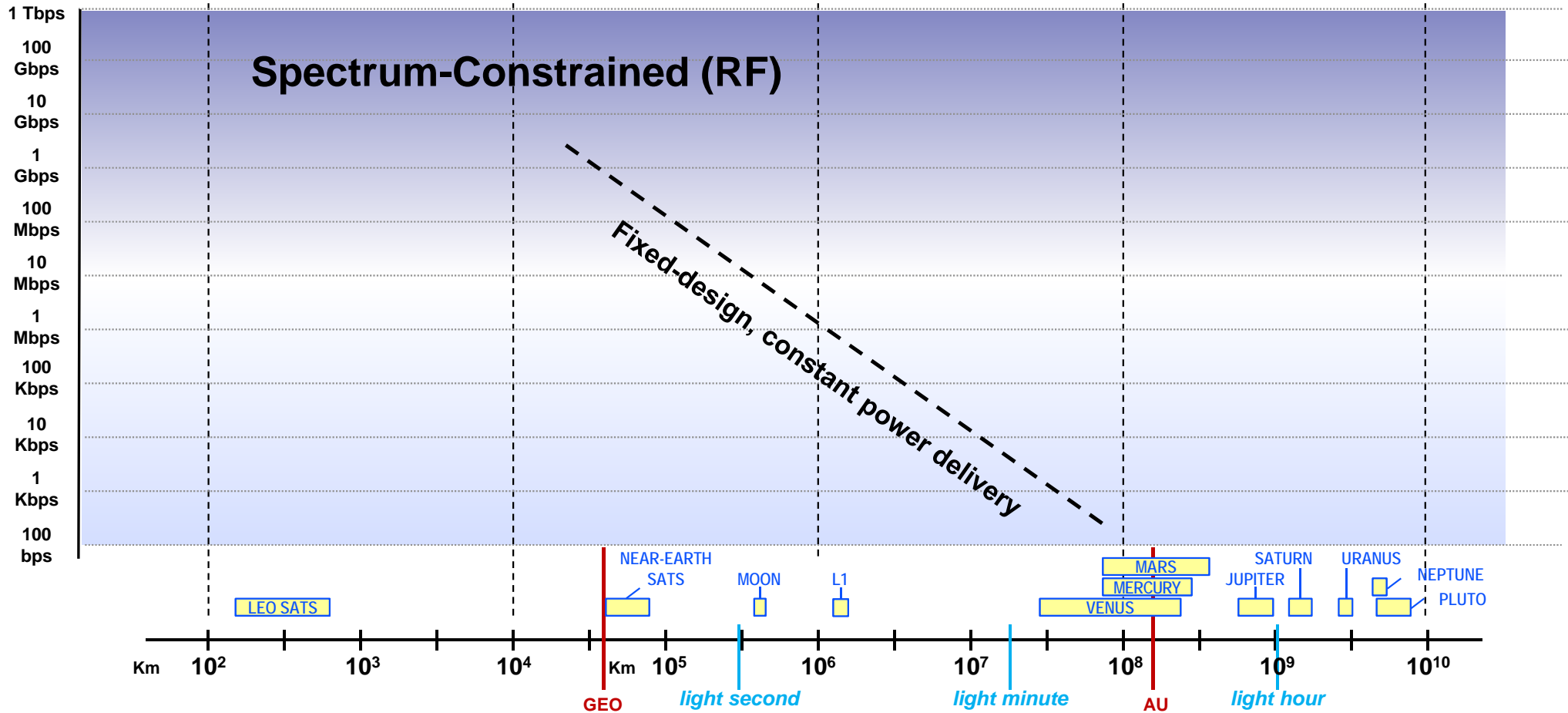


NASA Space Communication Challenges



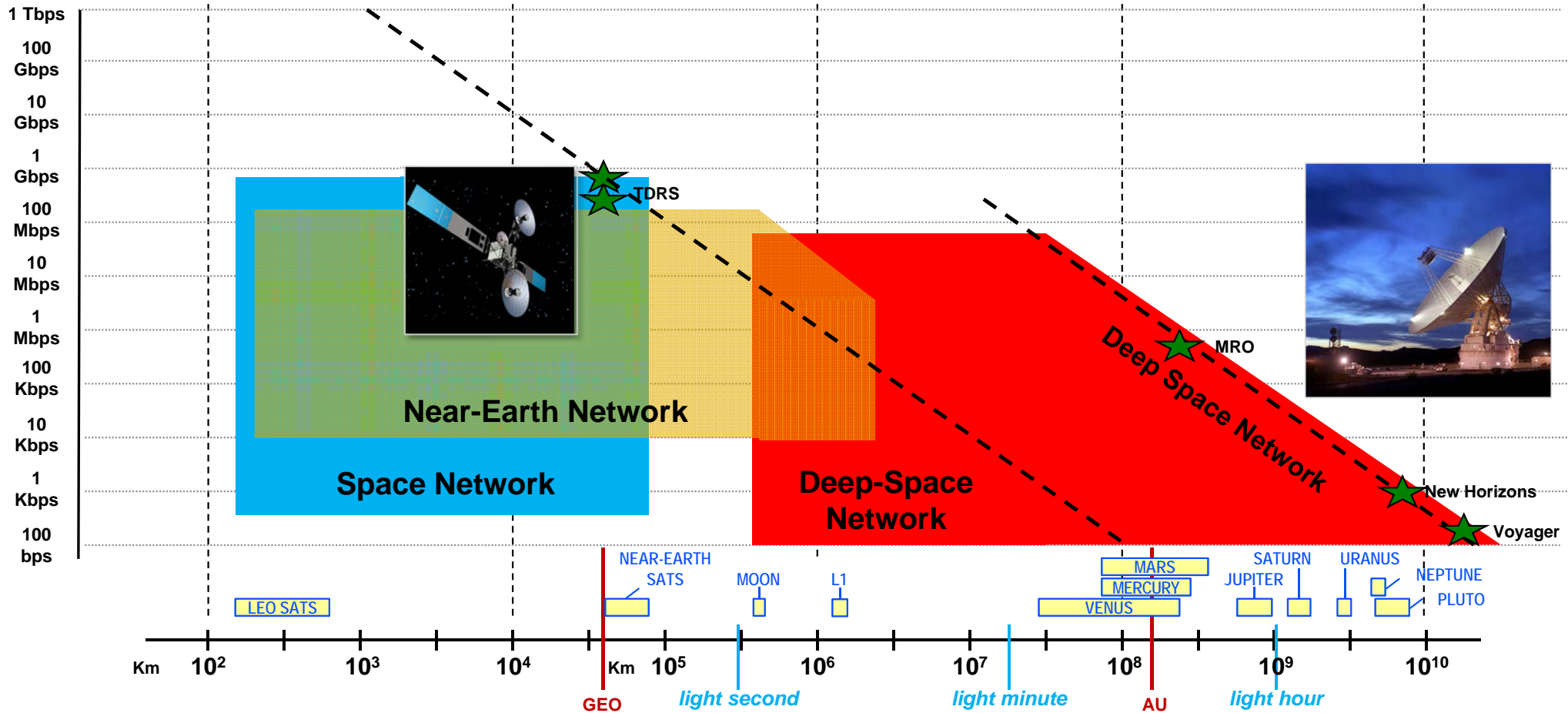
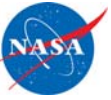


NASA Space Communication Challenges



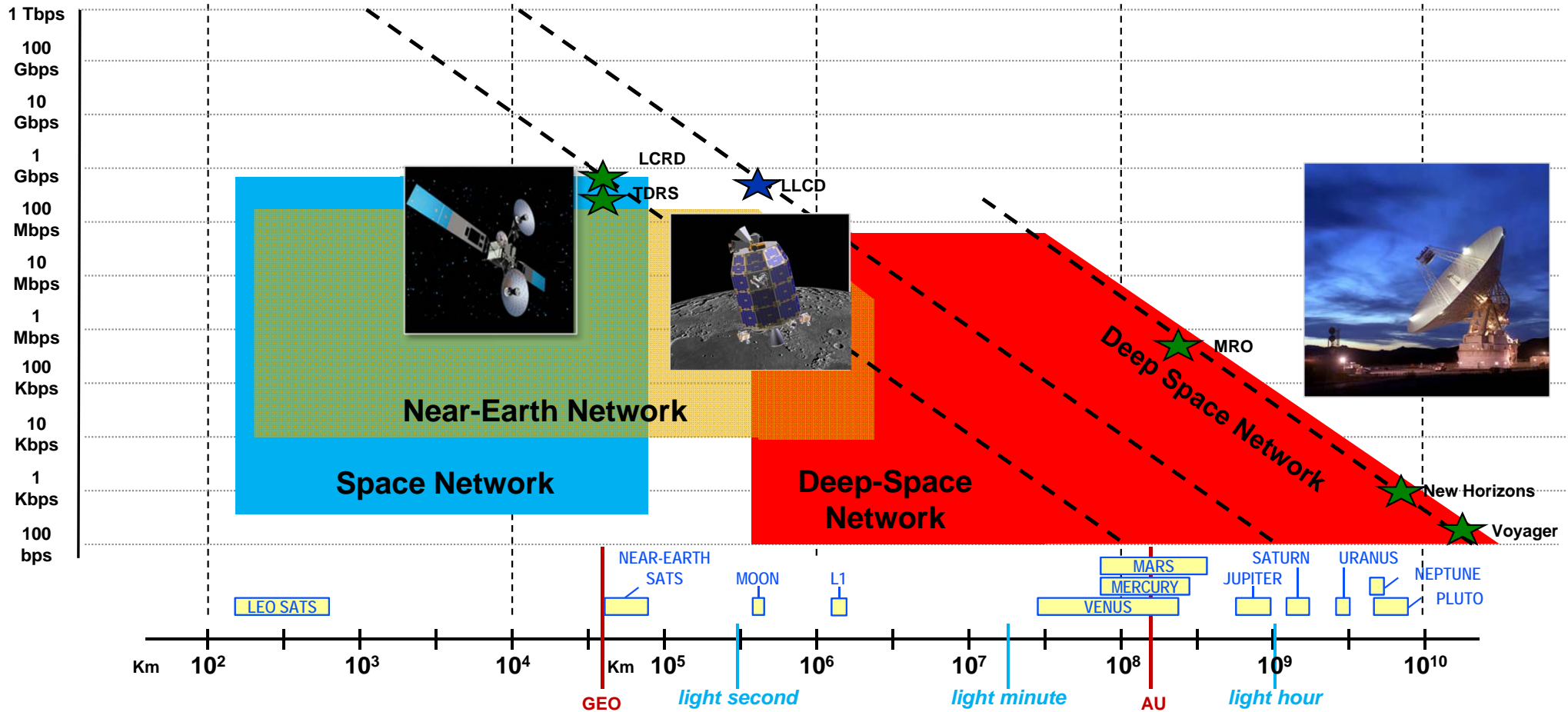


NASA Space Communication Challenges





NASA Space Communication Challenges





Lunar Laser Communication Demonstration

NASA's First Lasercom Demonstration (2013-2014)



Technology demo on NASA's Lunar Atmosphere and Dust Environment Explorer (LADEE)

- 622 Mbps downlink from moon
- 20 Mbps uplink to moon
- Longest lasercom link ever demonstrated

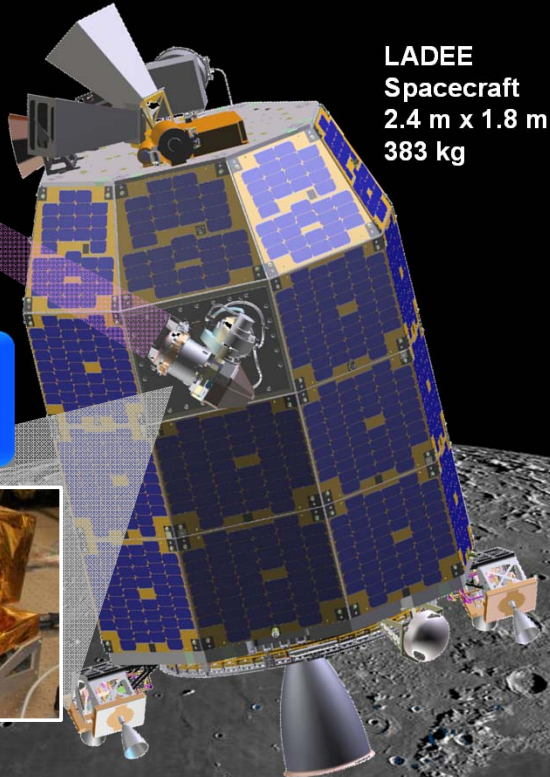
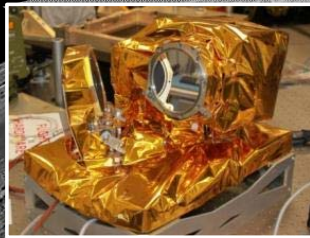


4 x 40-cm Collector
Photon-Counting Receiver



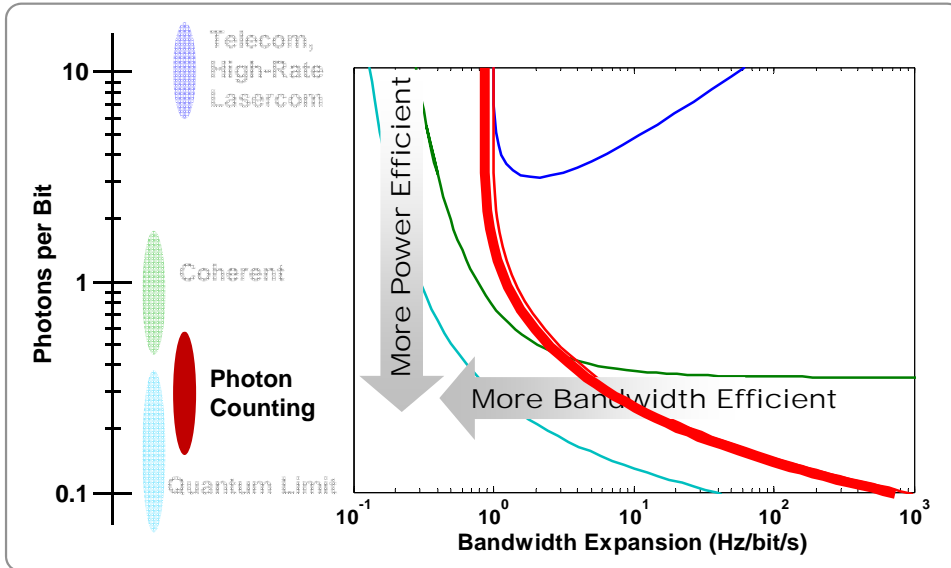
Lunar Lasercom
Operations Center
at MITLL

10-cm Transmit
Aperture
0.5-W Laser



LADEE
Spacecraft
2.4 m x 1.8 m
383 kg

LLCD Key System Parameters and Design Choices (1/4)



MIT-LL Geiger-Mode Avalanche Photodiode Arrays

~45 % detection efficiency at 1064 nm (1550 nm)

Speed:

- 350 ps timing jitter
- 1.6 μ s reset time

CMOS ROIC

Alumina Interposer

APD with μ -lens

MIT-LL Superconducting Nanowires

75 % detection efficiency at 1550 nm

30 ps jitter for single photons

3 ns reset time for 3 μ m square device

Radiation hard

Response from UV to Mid-IR

Closed-cycle Cryo-cooler

4-element SNSPD Array

Key System Parameters

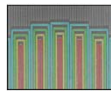
Receiver Selection

Modulation

Coding & Interleaving

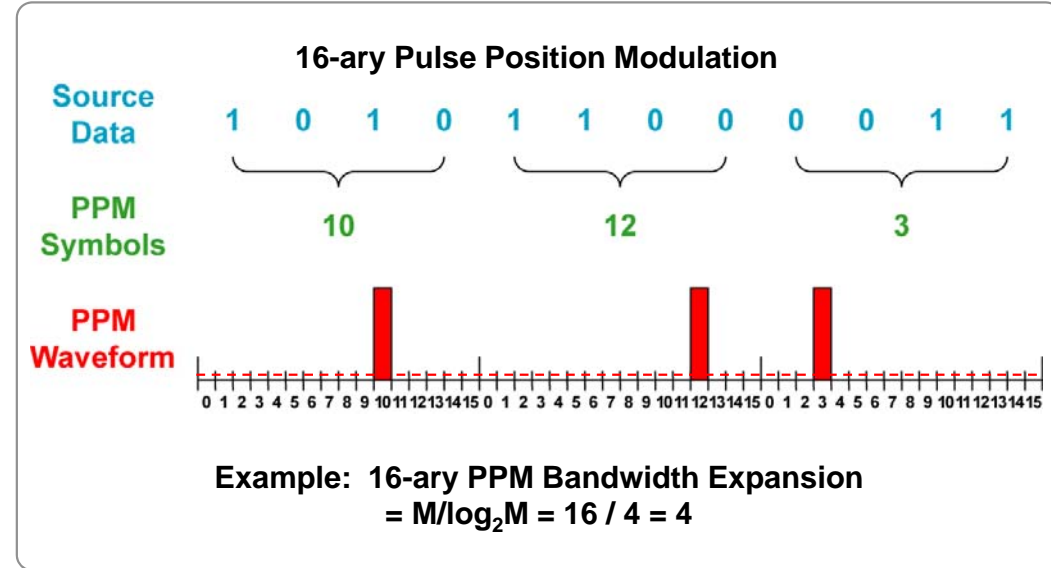
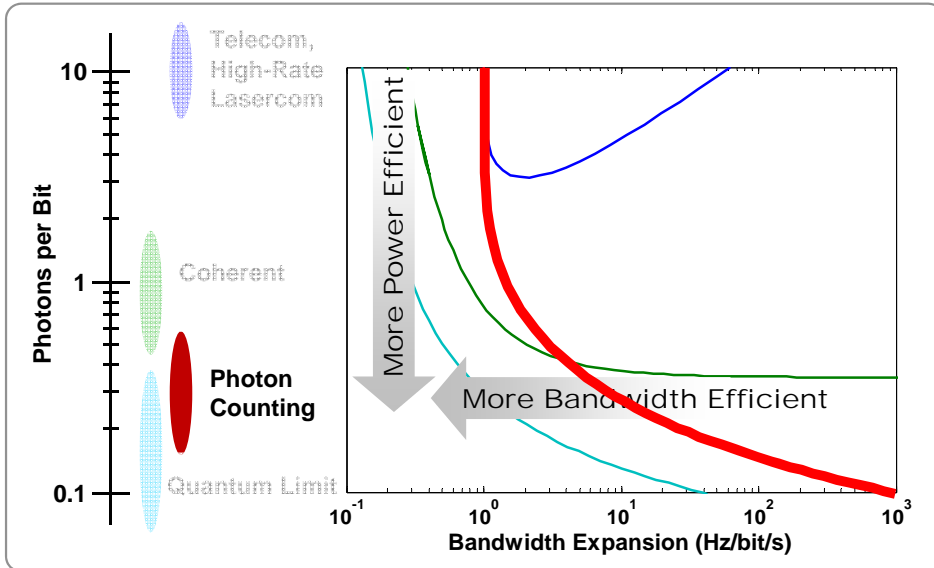
Ground Terminal Design

LLCD Design Choices



Superconducting Nanowire Arrays

LLCD Key System Parameters and Design Choices (2/4)



Key System Parameters

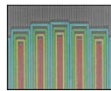
Receiver Selection

Modulation

Coding & Interleaving

Ground Terminal Design

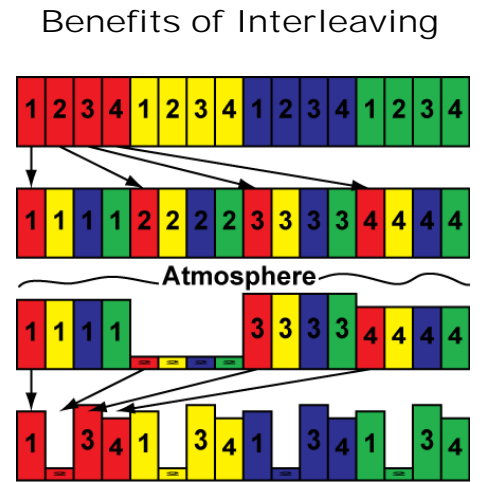
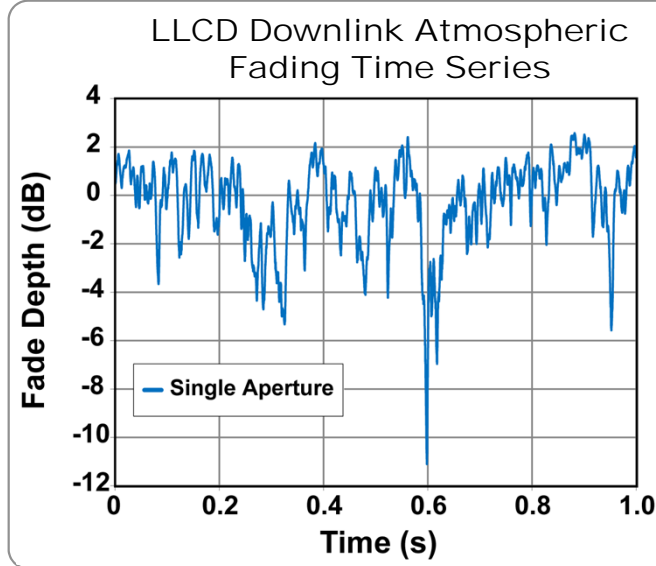
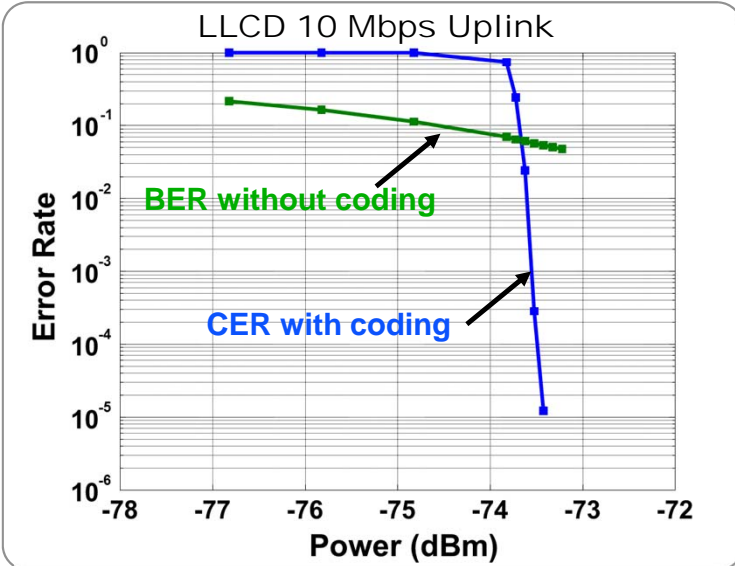
LLCD Design Choices



Superconducting Nanowire Arrays

- Pulse Position Modulation

LLCD Key System Parameters and Design Choices (3/4)



Key System Parameters

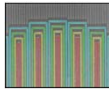
Receiver Selection

Modulation and Coding

Coding & Interleaving

Ground Terminal Design

LLCD Design Choices



Superconducting Nanowire Arrays

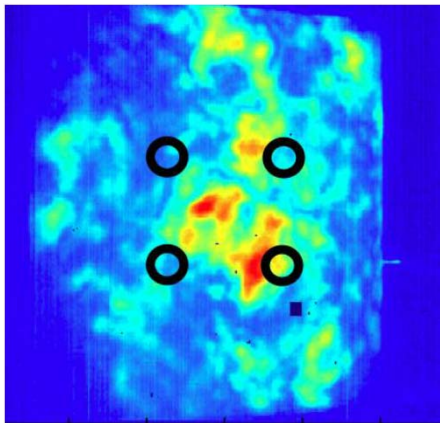
- Pulse Position Modulation

- $\frac{1}{2}$ rate Serially Concatenated Pulse Position Modulation (SCPPM) **JPL**
- 1 sec interleaver

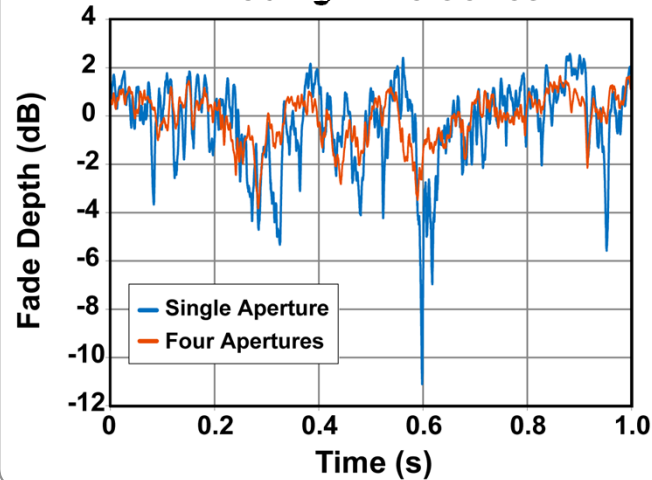
LLCD Key System Parameters and Design Choices (4/4)



Multiple Aperture Receiver
Spatial Diversity



LLCD Downlink Atmospheric
Fading Time Series



Ground-Based Telescope
Array (LLCD)

Key System
Parameters

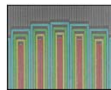
Receiver Selection

Modulation

Coding & Interleaver

Ground Terminal Design

LLCD Design
Choices



Superconducting
Nanowire Arrays

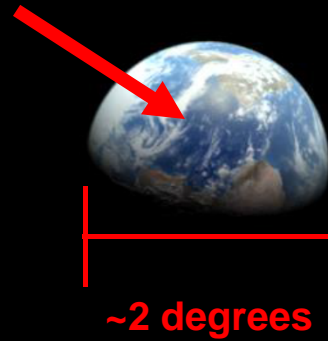
• Pulse Position
Modulation

• 1 sec interleaver
• Serially Concatenated
Pulse Position
Modulation (SCPPM) **JPL**

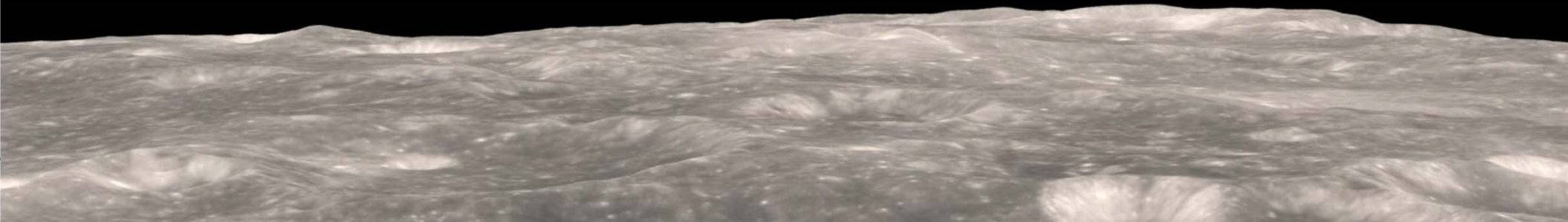
• Scalable
• Transportable
• Array 4 x 40 cm



Beam Size from Moon

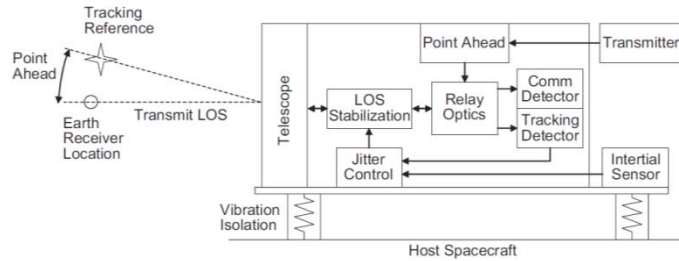


- 10-cm transmit aperture
- 15 μ rad beam
 - ~0.001 deg
 - ~6 km on Earth



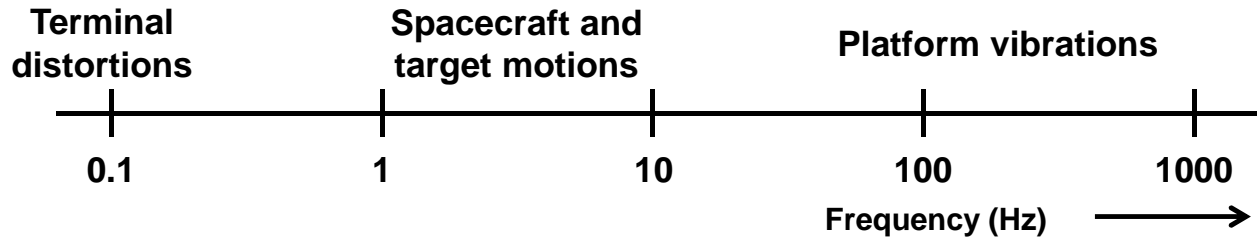


Beam Stabilization

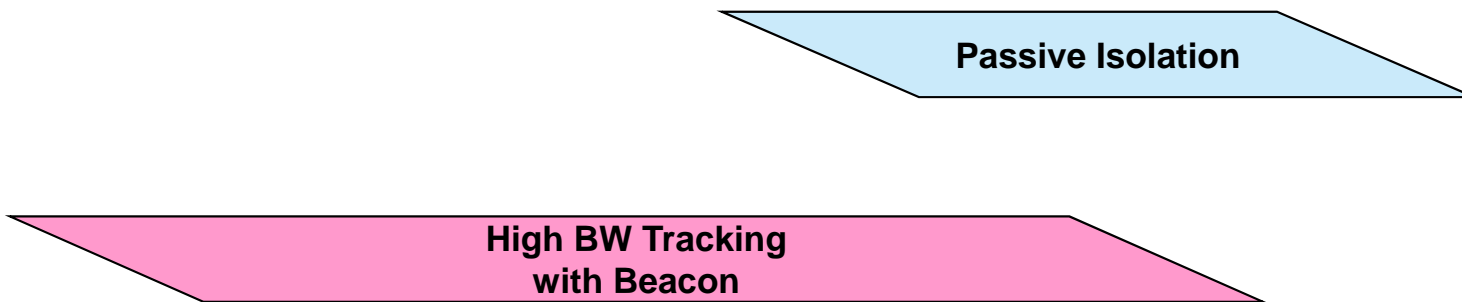


* Figure from "Deep Space Optical Communications", H. Hemmati, ed.

Disturbances



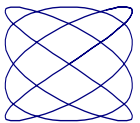
Stabilization Methods



Vibration Isolator

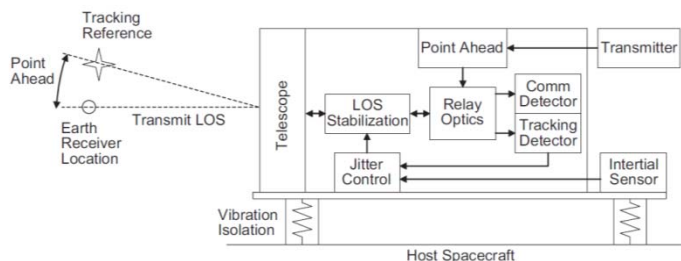


Electro-Optic Nutator with scan



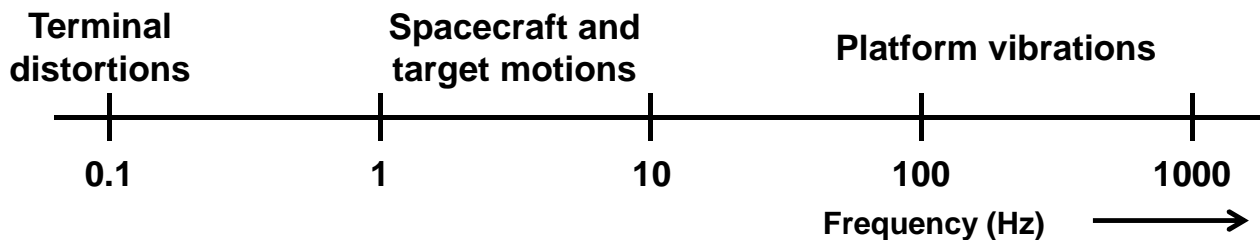


Beam Stabilization

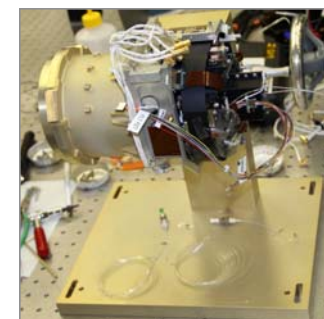
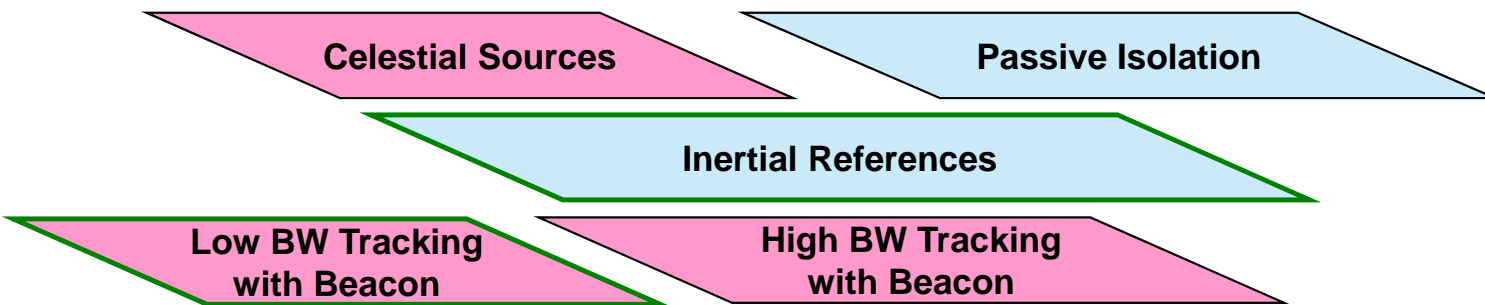


* Figure from "Deep Space Optical Communications", H. Hemmati, ed.

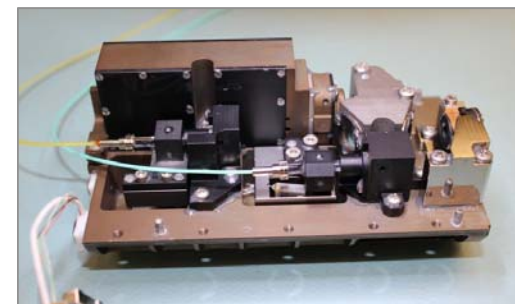
Disturbances



Stabilization Methods



Inertially-Stabilized Terminal

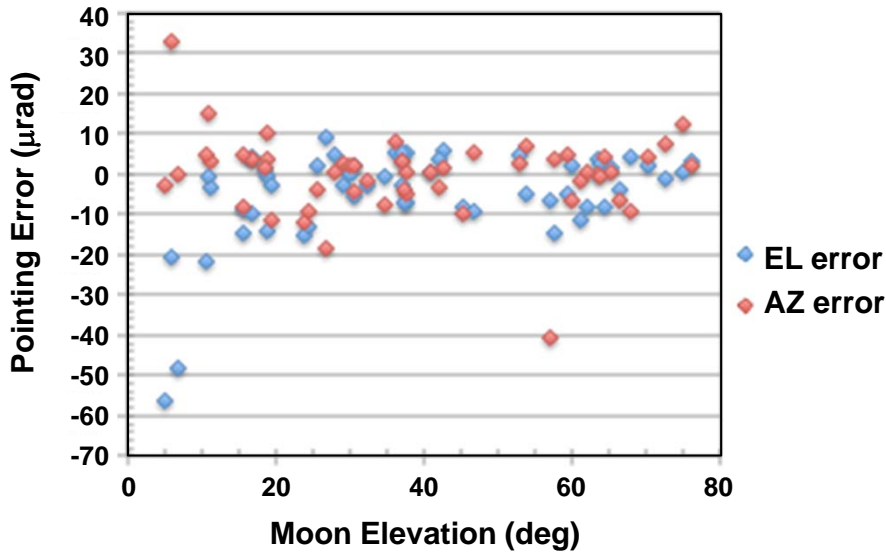


Small Optics with Piezo Nutator

Acquisition and Downlink Communication Performance

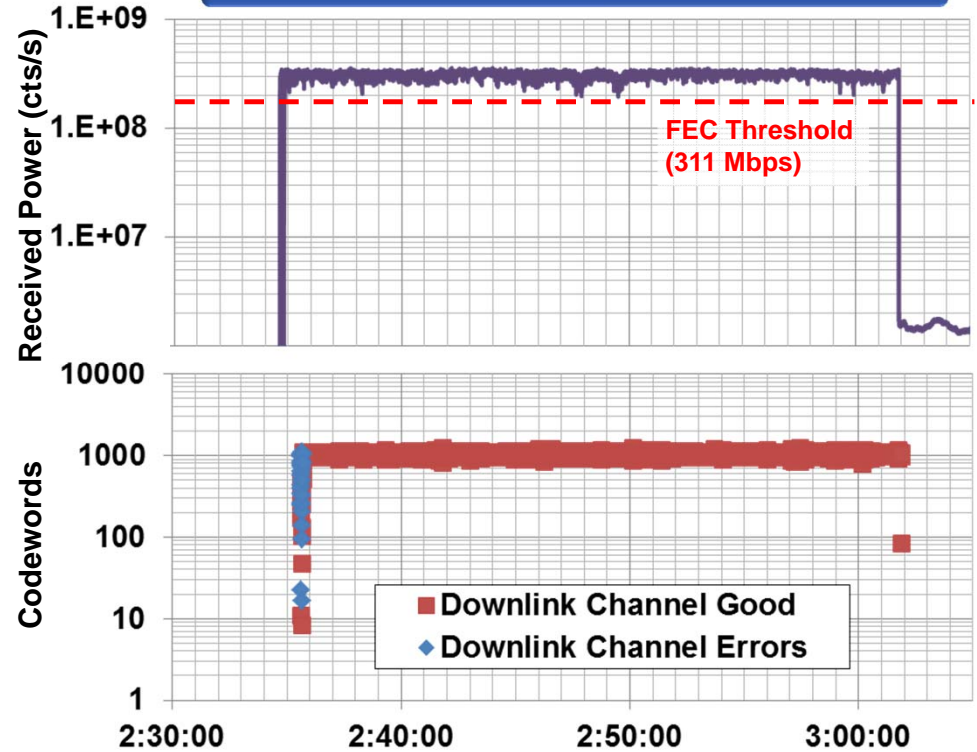


LLGT Initial Pointing Error at Acquisition



2-axis pointing error for elevation > 7 deg = 11.2 μrad rms

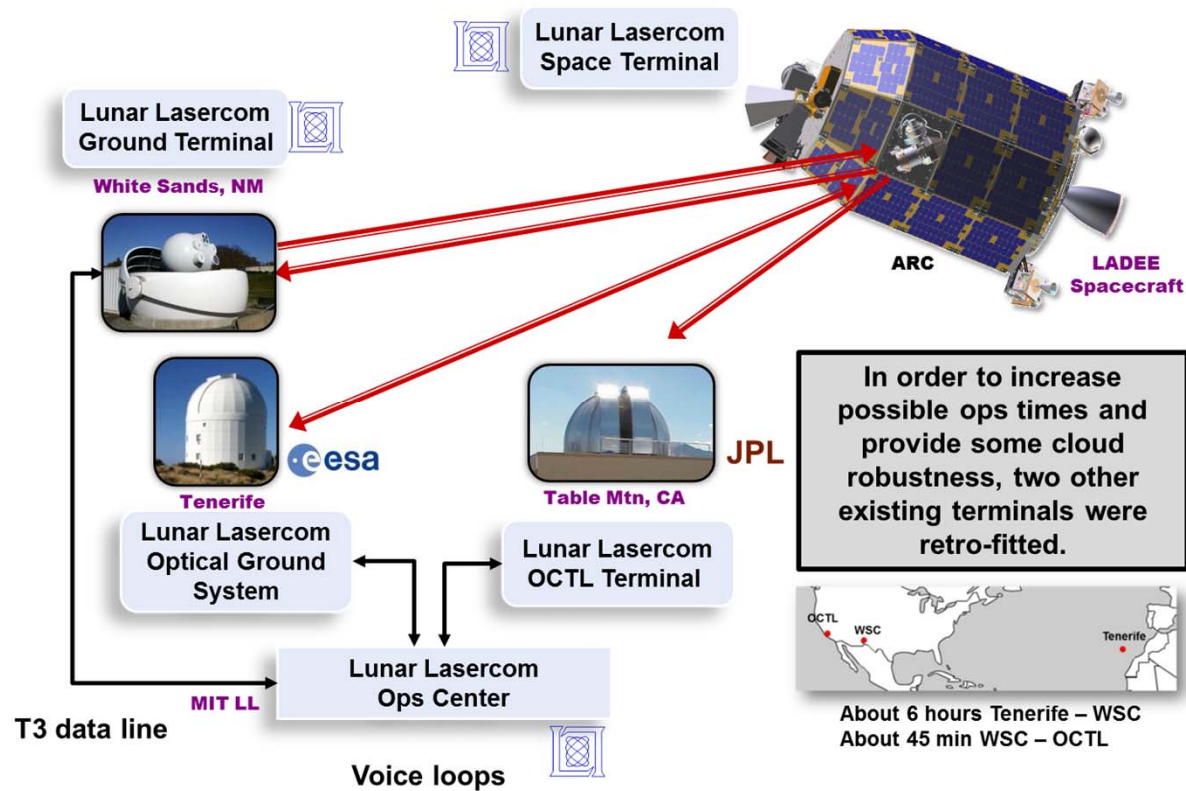
Downlink Comm Performance



Error-free communication achieved



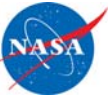
LLCD System



International community understands how to build interoperable ground terminals



Lasercom Operations in the Presence of Clouds



Thin Cirrus Clouds

- **Reduce data rate**
 - Downlink data rates from 39 to 622 Mbps
 - Uplink data rates 10/20 Mbps



Intermittent Clouds

- **Mid-pass handover between ground stations**
- **Delay / disruption tolerant network**



Thick Cloud Cover

- **Use alternate ground station!**

LLCD Oct–Nov Ops

Day	Block 1				Block 2				Block 3				Block 4		
	1	2	3	4	1	2	3	4	1	2	3	4	1	2	3
1	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
2	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
3	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
4	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
5	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
6	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
7	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
8	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█
9	█	█	█	█	█	█	█	█	█	█	█	█	█	█	█

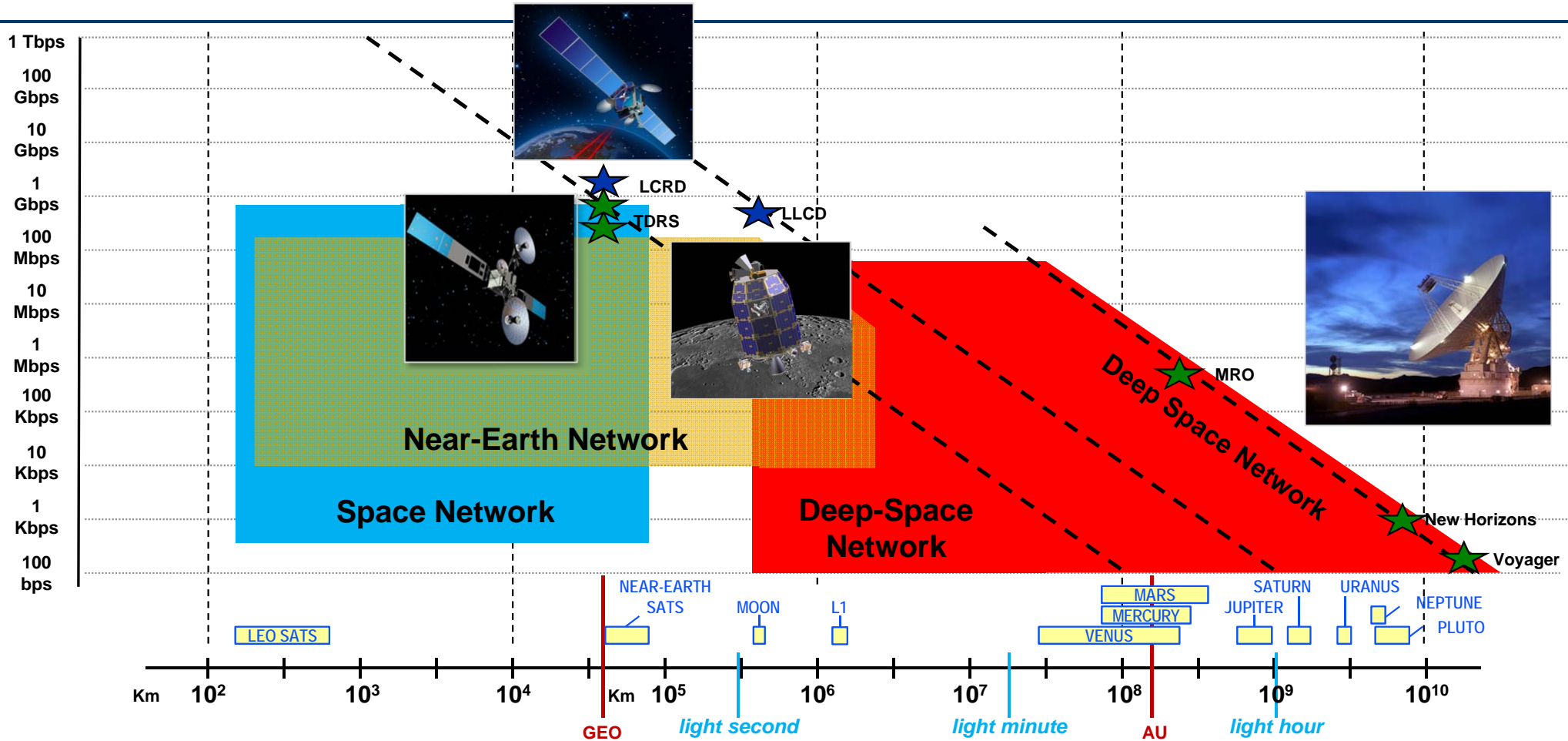
- █ Operations
- █ Clouded out
- Not available

- **10% passes clouded out**
- **11% alternate ground station used**

All solutions were demonstrated during LLCD

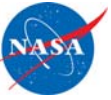


NASA Space Communication Challenges

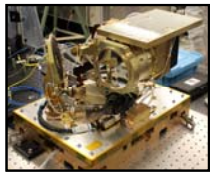




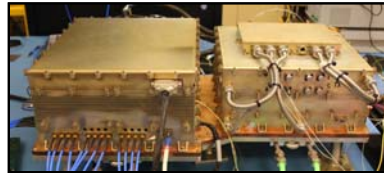
Space Lasercom Operational Pathfinder



Qualified Subsystems being Integrated at MIT-LL

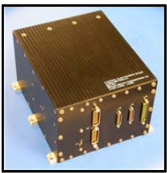


Optical Module

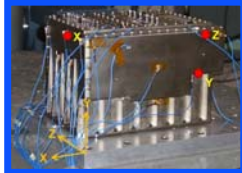


Multi-Rate Modem

Qualified Subsystems being Procured from Industry



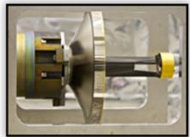
Pointing Processor (Moog BRE)



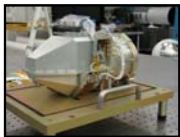
Digital Processor (SEAKR Eng., Inc)



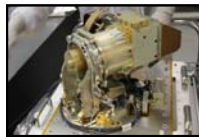
Modem Electronics (Aeroflex Cobham)



Optical Subassembly (ITT-Exelis)



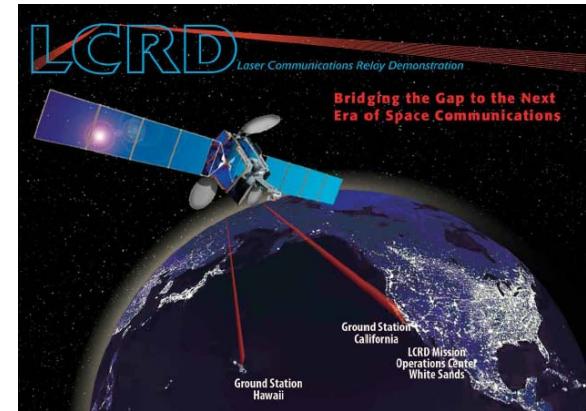
Inertially-Stable Platform (ATA)



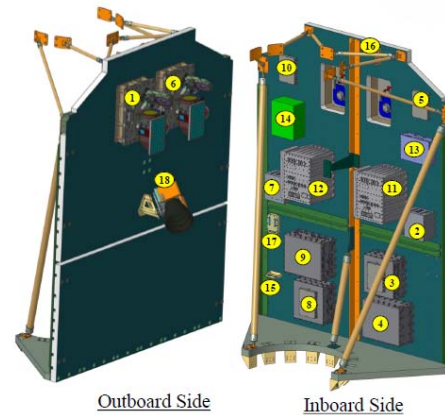
Gimbal & Latch (Sierra Nevada Corp)



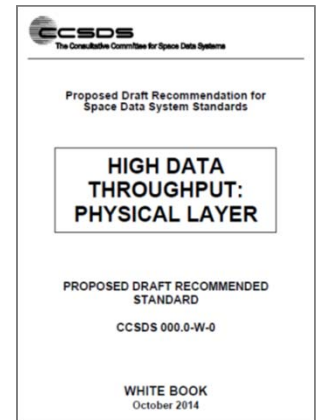
Solar Window Assembly (L3-SSG)



Payload Configuration

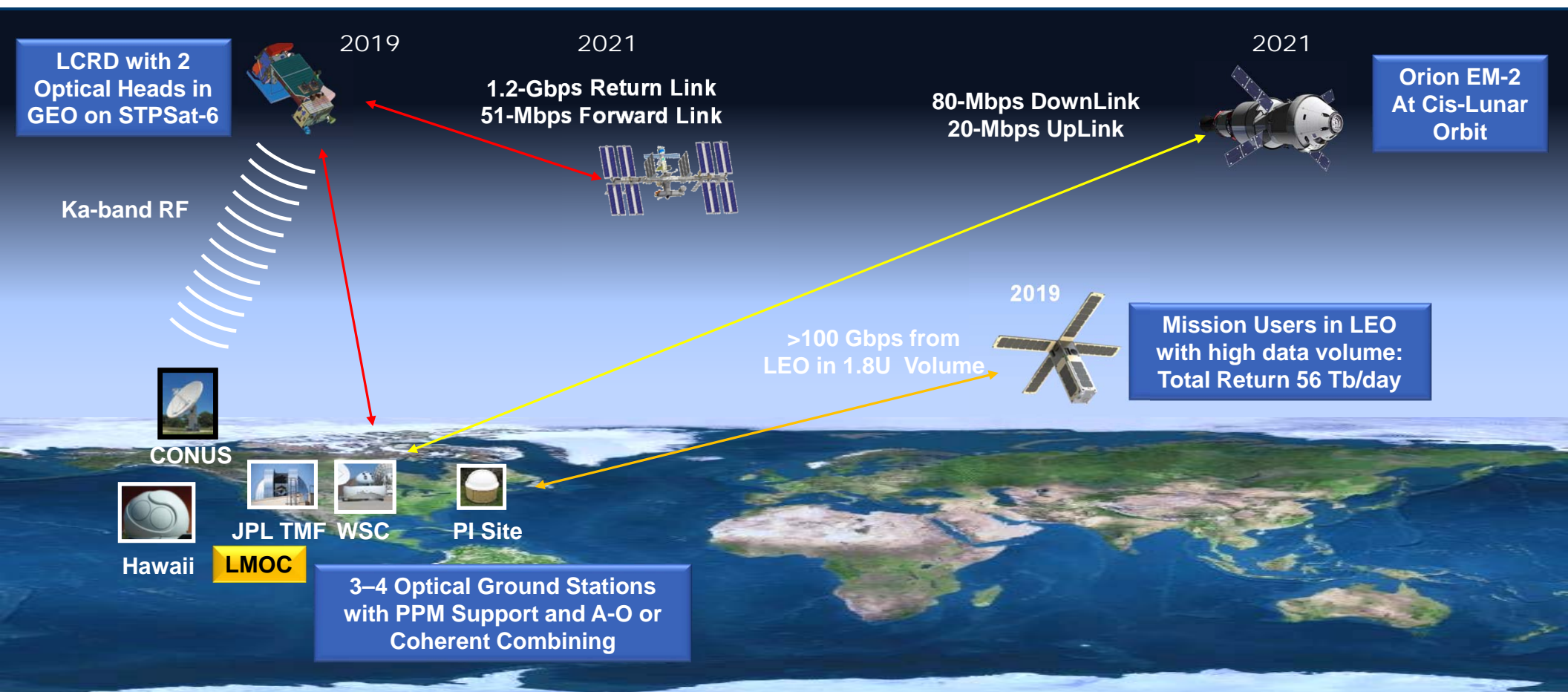


International Standards



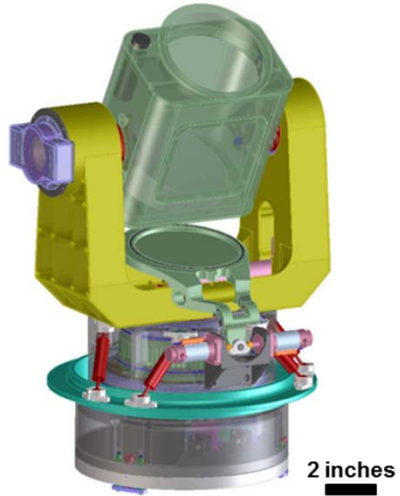
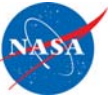


NASA's Next Generation Near-Earth Space Comm and Nav Network





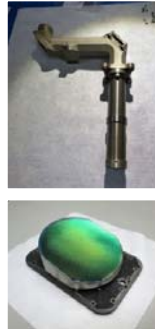
Next Generation Terminal (NGT) Optical Module



NextGen Terminal (NGT)
 estimated 13 kg mass

- Modular
- Scalable
- Leverages heritage programs

MODULAR



Telescope and Relay



Latch and Gimbals

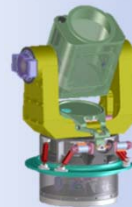


Optical Bench

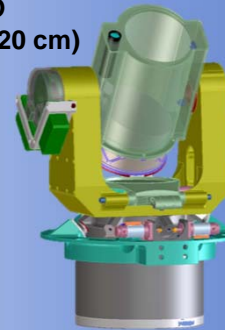


SCALABLE

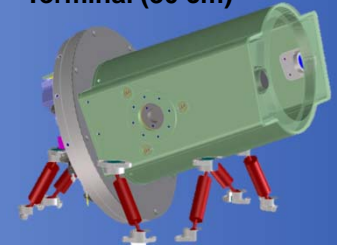
LEO User Terminal (10 cm)



GEO Terminal (20 cm)



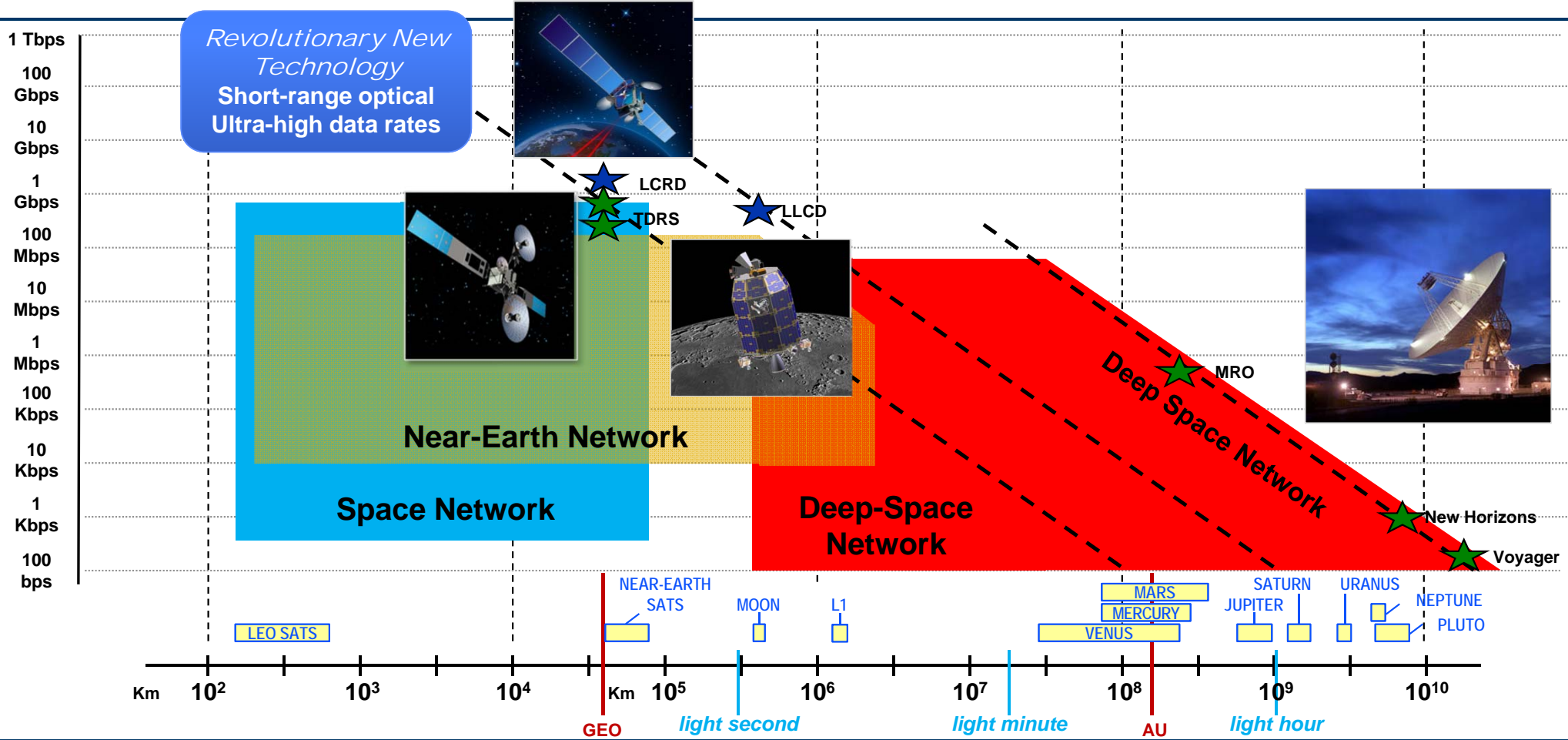
Deep Space Terminal (30 cm)



Subassemblies for initial NGT prototype are being developed by industry



NASA Space Communication Challenges





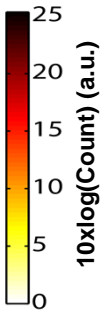
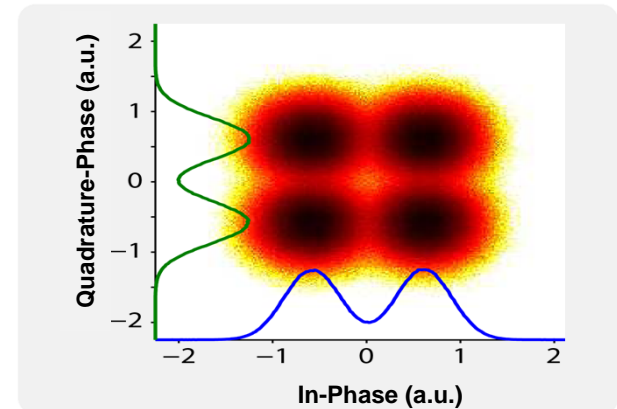
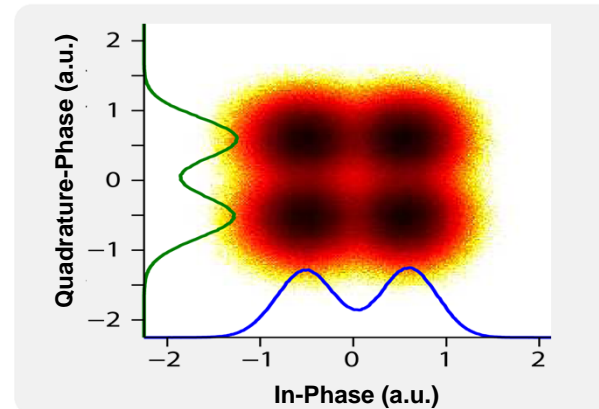
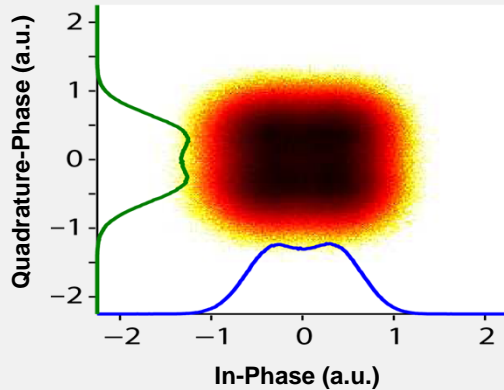
Coherent Optical Receiver Technology



25 Gbps QPSK
25 GHz Symbol Clock
Rate-1/2 FEC

40 Gbps QPSK
25 GHz Symbol Clock
Rate-4/5 FEC

45 Gbps QPSK
25 GHz Symbol Clock
Rate-9/10 FEC



- High-sensitivity scalable multi-aperture receivers
- High data-rate communications leveraging fiber-optic communications technology

MIT-LL is developing high-rate efficient coherent modem technologies for lasercom applications



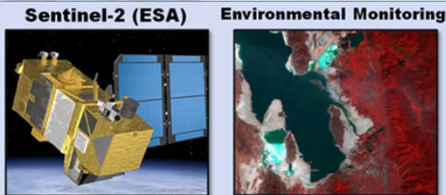
Low-Earth Orbit Direct-to-Earth Optical Links



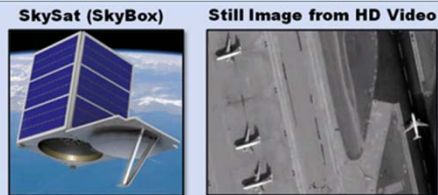
Transmitter
0.5 W,
2-cm Telescope

- Large data volume – many Terabytes per day!
- Small space terminal – CubeSat scale!
- Small, low-cost ground terminal – widely deployable!

200 Gbps Downlink from LEO
can deliver 7 Terabytes per day
to single ground station



- Sentinel-2 (ESA) Environmental Monitoring**
- Multispectral imaging, 10-60m resolution
 - Generates ~450 Mbps (compressed)
 - Downlinks ~0.5 Terabytes per day via geosynchronous relay
 - Continuous operation would generate ~4.9 Terabytes per day



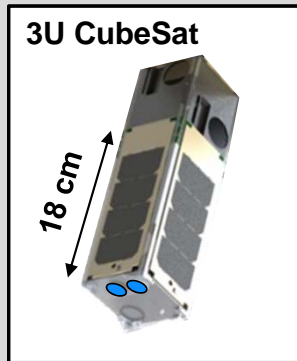
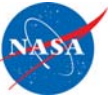
- SkySat (SkyBox) Still Image from HD Video**
- 1080p HD video, 1m resolution
 - Generates ~100 Mbps (compressed)
 - Downlinks ~0.2 Terabytes per day direct to Earth
 - Continuous operation would generate ~1.1 Terabytes per day



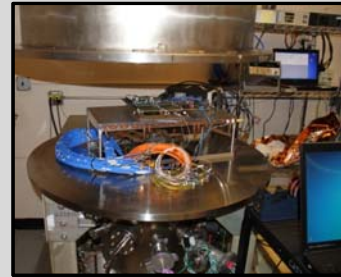
Receiver
40-cm
Telescope



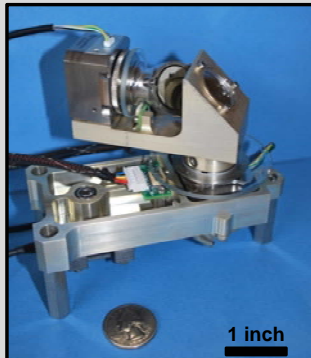
DTE Risk Reduction Efforts



- Concept for 1.8U lasercom payload being developed with NASA SCaN
 - Working with NASA STMD to procure a CubeSat



- Space qualification of COTS components
 - Vacuum
 - Thermal
 - Vibration
 - Radiation



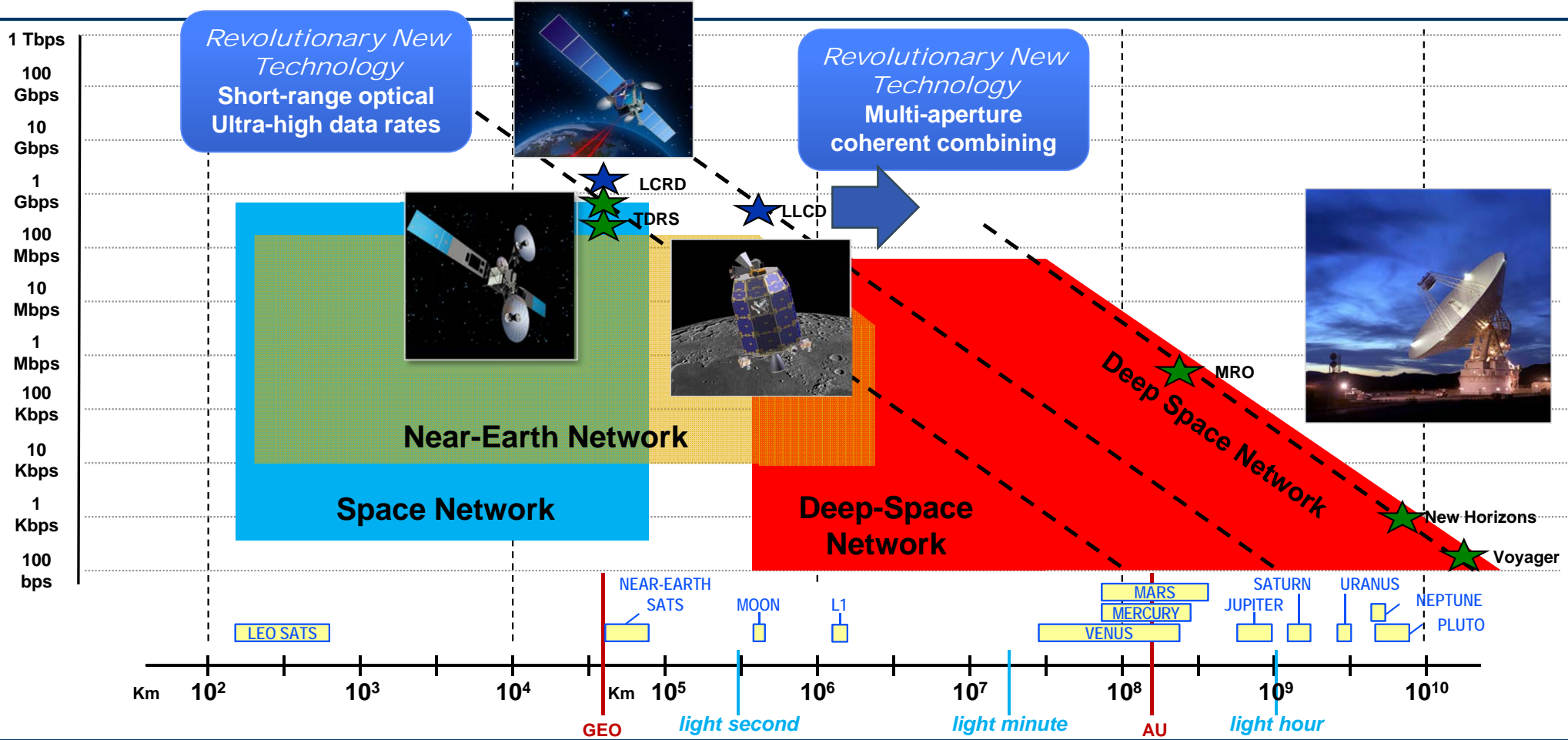
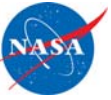
- Compact Beam Director
 - 1-2 cm aperture
 - 0.5 kg
 - Hemispherical field of regard



- End-to-end comm testbed
 - Fading channel emulator
 - High-bandwidth data buffer



NASA Space Communication Challenges





Large Scientific-Class Ground Terminals

- Historically require fixed-site high-cost installations
- Typically utilize large >1-m telescope apertures
 - Coupled via coudé path to laboratory class optical tables
- Examples include:
 - JPL's Mount Palomar Hale 5.1-m Telescope
 - JPL's 1-m Optical Communication Telescope Laboratory (OCTL)
 - ESA's 1-m Teide Observatory in Tenerife, Spain



Mt. Palomar 5.1-m Hale Telescope



Table Mountain 1-m OCTL Telescope

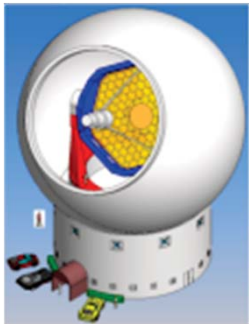


Teide Observatory 1-m Telescope



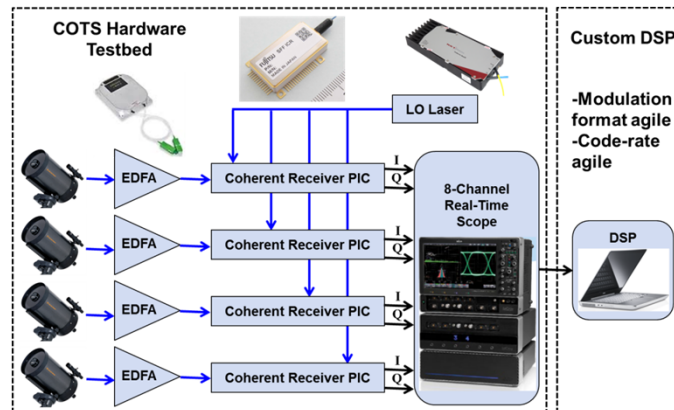
Next-Generation Optical Gateway Concepts

- Which solution is best for large area optical ground stations?
 - Cost for equivalent performance
 - Daytime operation and pointing near the sun



NASA Integrated Telescope

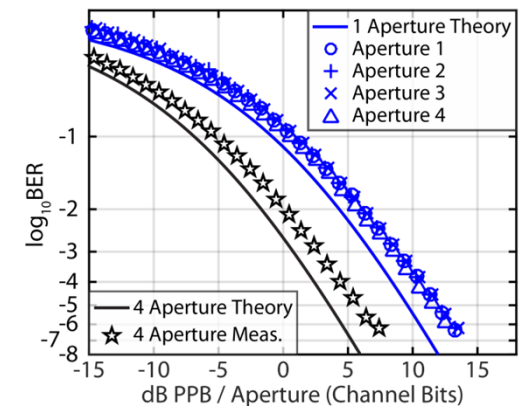
- 12-m aperture diameter
- ~100 Hexagonal segmented mirrors
- Dedicated for optical communications



Coherent Multi-Aperture Array

- Leverage COTS telescopes and telecom components
- Develop custom DSP algorithms to support multi-aperture arrays for next generation optical ground terminals

Lossless Coherent Combining





Summary

Findings:

- **Lasercom offers higher rates and reduced SWaP for space communications**
 - Requires cloud-free line-of-sight
- **Lasercom has been matured to at least TRL-6 and is ready for transition to operations**
- **Space-to-air/space and space-to-ground has been matured to TRL-6 or beyond**
- **Lasercom is an enabling technology for other important links**
 - Air-to-air, LEO and deep-space

Recommendations:

- **Insert lasercom into program-of-record, e.g. relay**
- **Invest in manufacturability**
- **Continue R&D on other links**



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